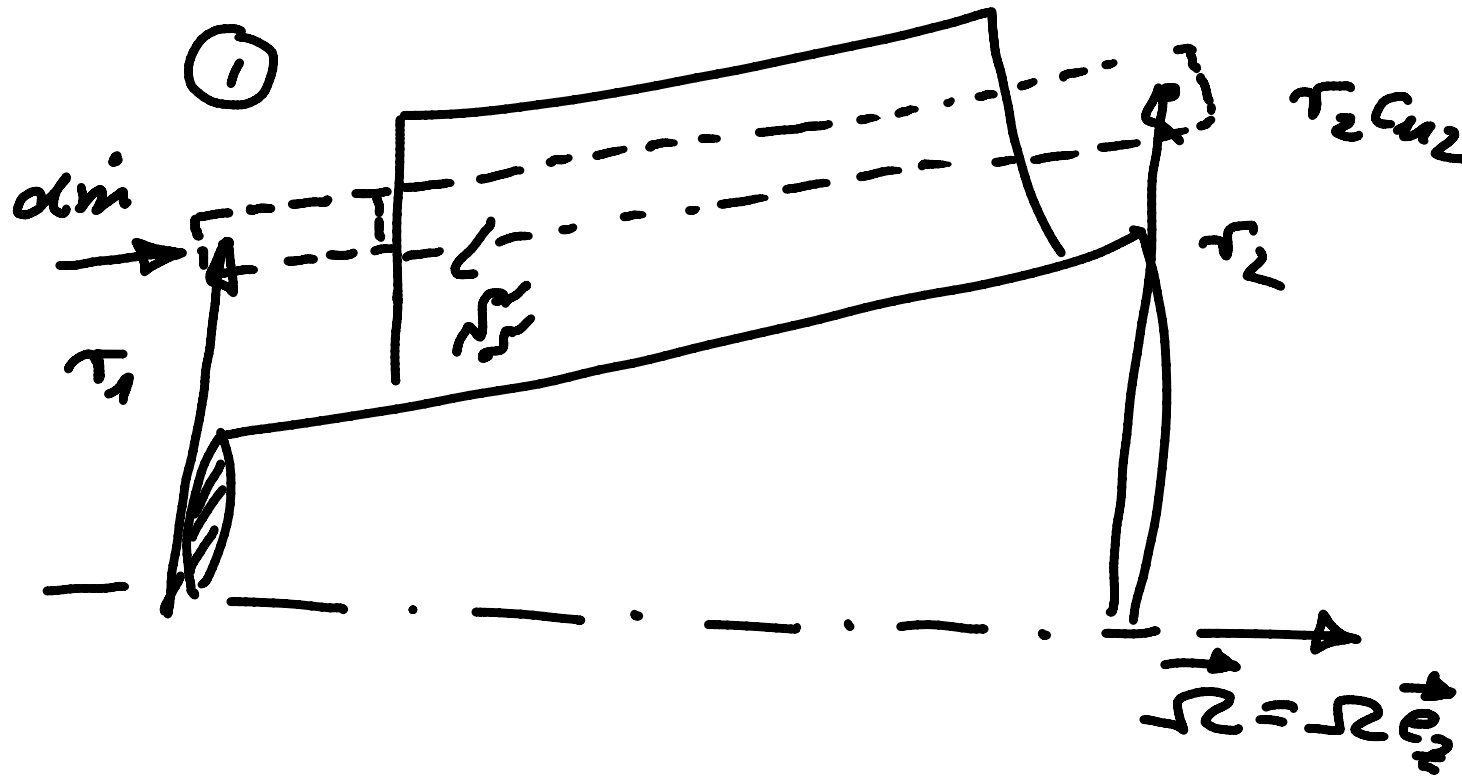


Euler's Turbinengleichung

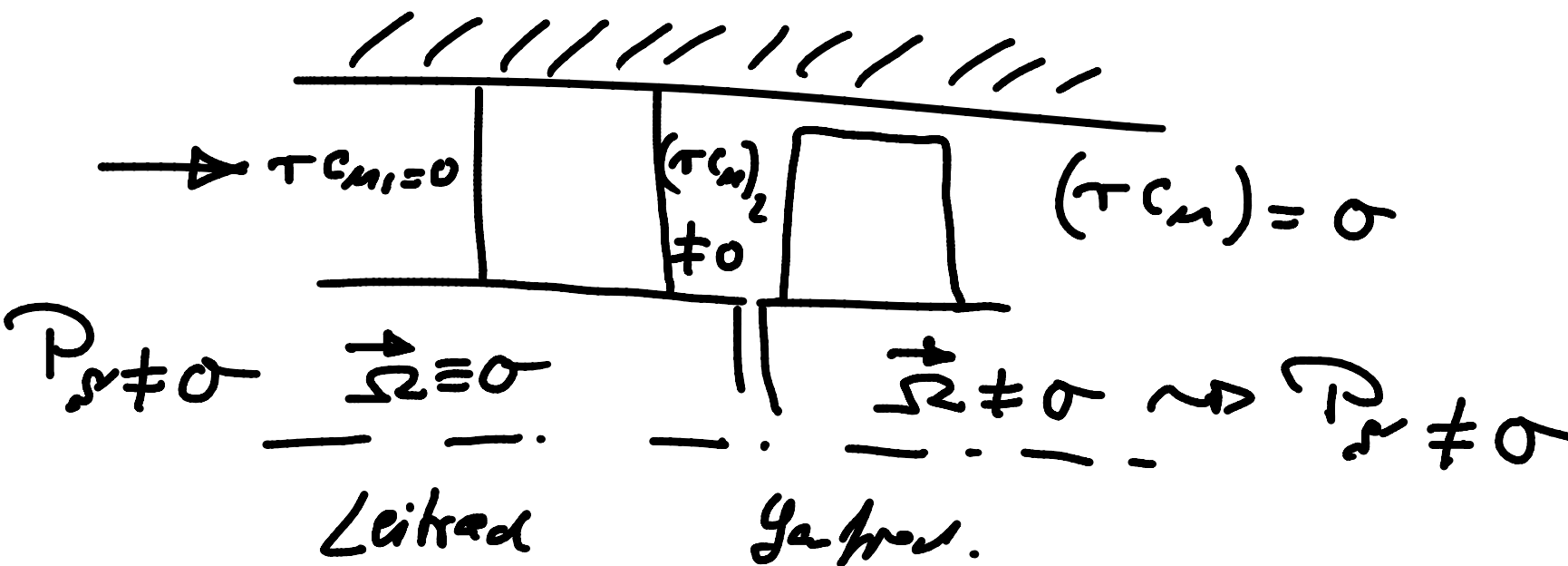
$$\frac{dW_{sz}}{dn} = r_2 c_{u2} - r_1 c_{u1} \quad (2)$$

gilt für
für Injektoren (I)
aber auch für
den Gegentrieb. (B)





Axiale Durchströmung.



Drehfrei.

Drehl.

Drehfrei.



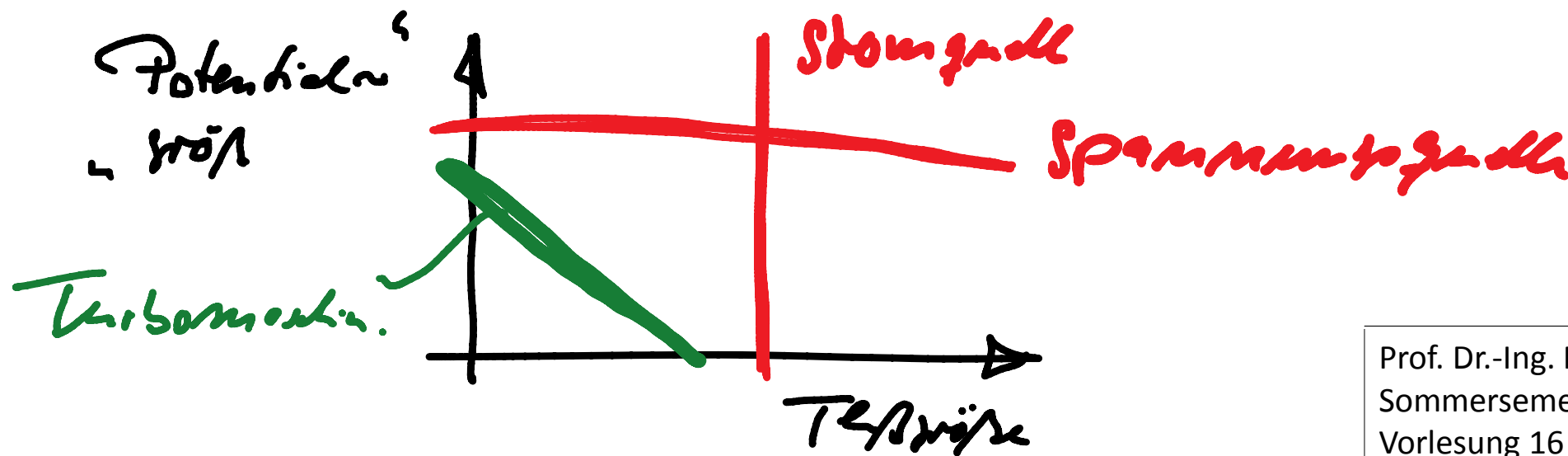
Euler's Turbine equation $\cdot \vec{\Omega} = \vec{H}$

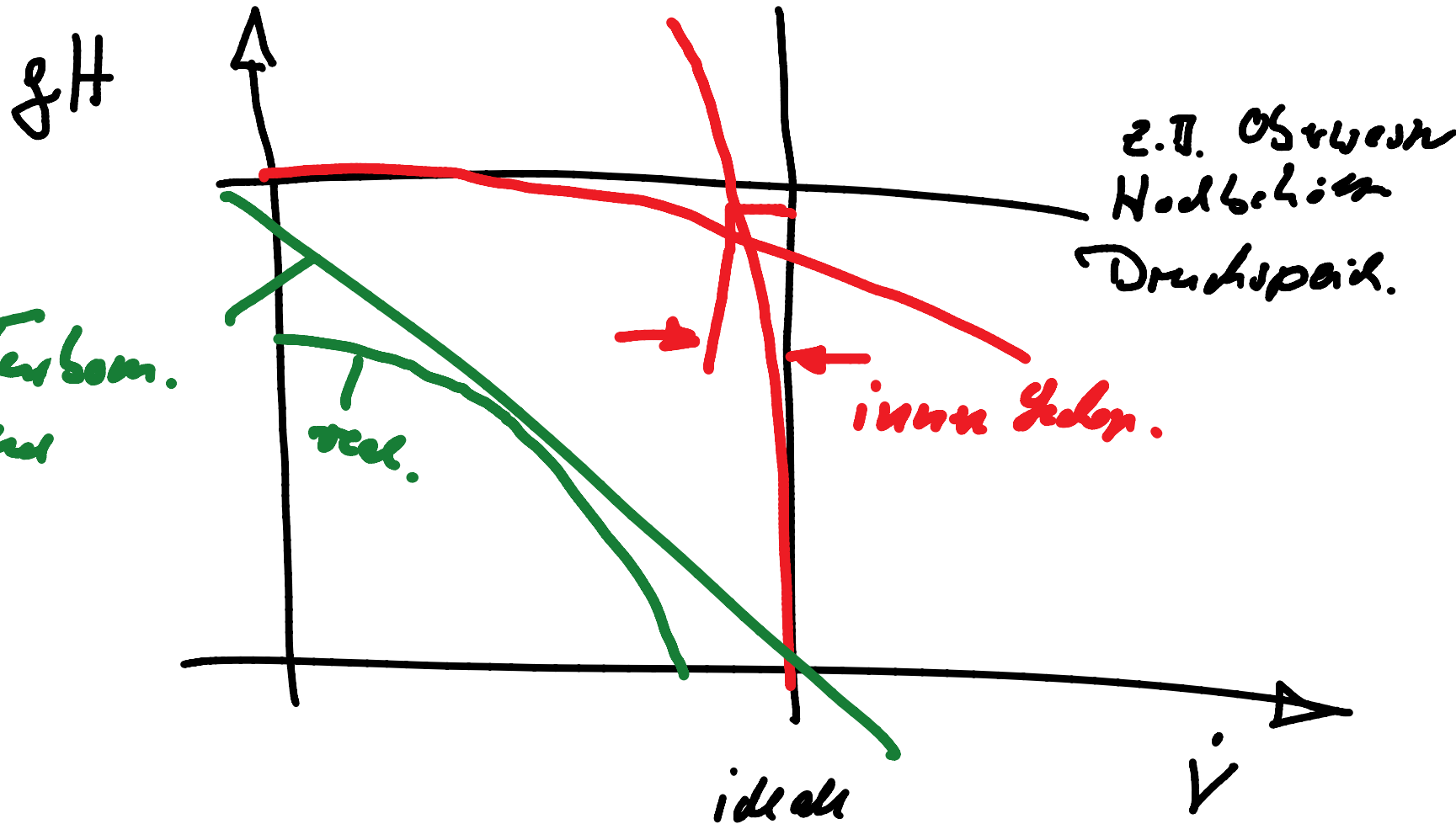
Energy equation \leadsto Euler's energy line
line Turbomachine.

Druckänderung = $\int \frac{1}{\rho} (\text{Volumenstrom})$

$\hat{=}$ Potentialenergie

$\hat{=}$ elektrisch Span.

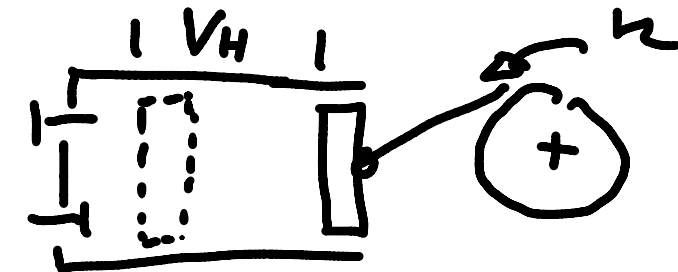




ideale
Verdrängungs-
maschine.

$$\dot{V} = n V_H z$$

V_H Hubvolumen
Schubvolumen
 n Drehzahl
 z Zahl der Arbeitskammern



1. Hauptsatz für $\dot{Q} \equiv 0$ (adiabot)

$$\overline{\frac{d}{dt}} \equiv \sigma$$

$$\frac{dP_{\text{st}}}{d\ln} = h_{t2} - h_{t1} = \frac{1}{\gamma \pm 1} gH$$

+1 Arbeitsmot.
-1 Wellmot.

$$gH := C_2 - C_1 = P_2 - P_1 + \psi_2 - \psi_1 + \frac{c_2^2}{2} + \frac{c_1^2}{2}$$

$$P = \int \frac{dP}{\rho}$$



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Sommersemester 2012
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2. Exakte Turbinengleichung $\bullet \vec{\Omega} = \Omega \vec{e}_z$

$$\frac{dM_z}{dt} \Omega = T_2 \Omega c_{m2} - T_1 \Omega c_{m1}$$

$$\frac{dP_{\text{rot}}}{dt} = M_2 c_{m2} - M_1 c_{m1}$$

$$\underbrace{h_{t2} - h_{t1}} = \underbrace{M_2 c_{m2} - M_1 c_{m1}}$$

energetisch Größe.

rein kinematisch Größe.



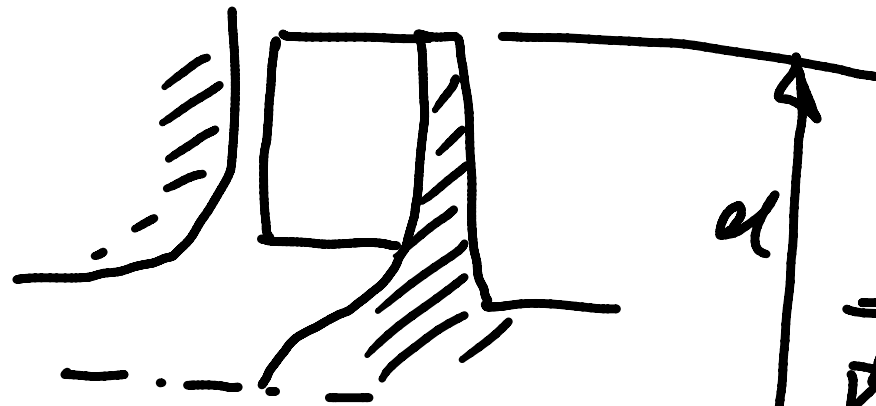
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mit $h_{t2} - h_{t1} = \frac{1}{z^{\pm 1}} g H$

$$g H = z^{\pm 1} (u_2 c_{u2} - u_1 c_{u1}) \quad \Bigg| \quad \frac{1}{u_2^2/2}$$



$$u_2 = \Omega r_2 = 2\pi n r_2 = \pi n a$$

Wichtiges Gesetz.



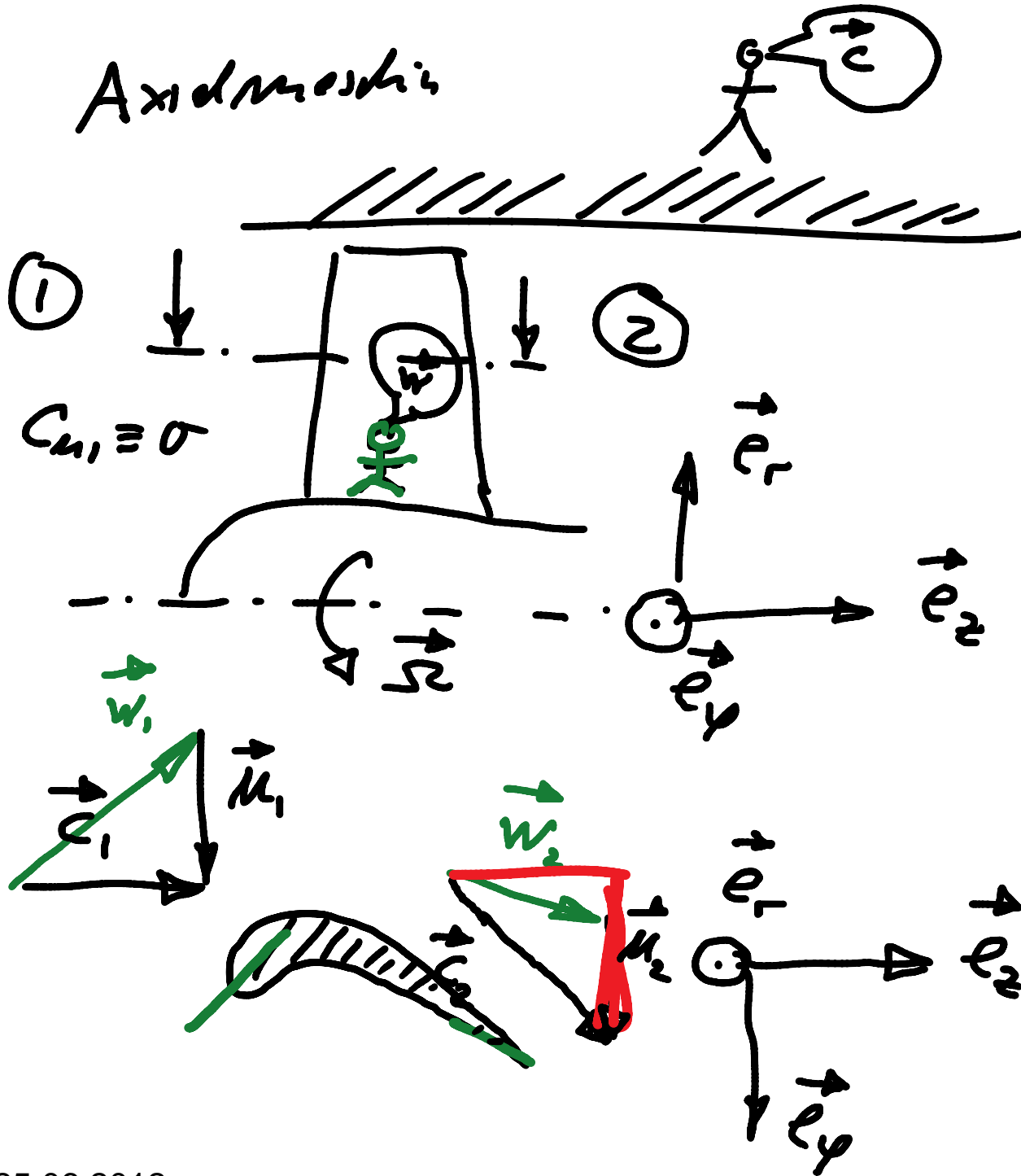
$$\left(\frac{gH}{u_2^2/2} \right) = z^{\pm 1} \left(2 \frac{C_{u2}}{u_2} - 2 \frac{T_1}{T_2} \frac{C_{u1}}{u_2} \right)$$

$\stackrel{!}{=} \Psi$ Druckkoeff.

$\frac{C_{u1}}{u_2}$ Verdreh $\equiv 0$ für ein dreifach Abspr.



Axidmaschi



Geschwindigkeitsdreieck

$$\vec{c} = \vec{w} + \vec{u} + \vec{v}$$

- \vec{c} absolute Geschw.
- \vec{w} relativ Geschw.
- \vec{u} Umfangsgeschw.
- \vec{v} Führungsgeschw. $\equiv 0$ i.d.R.

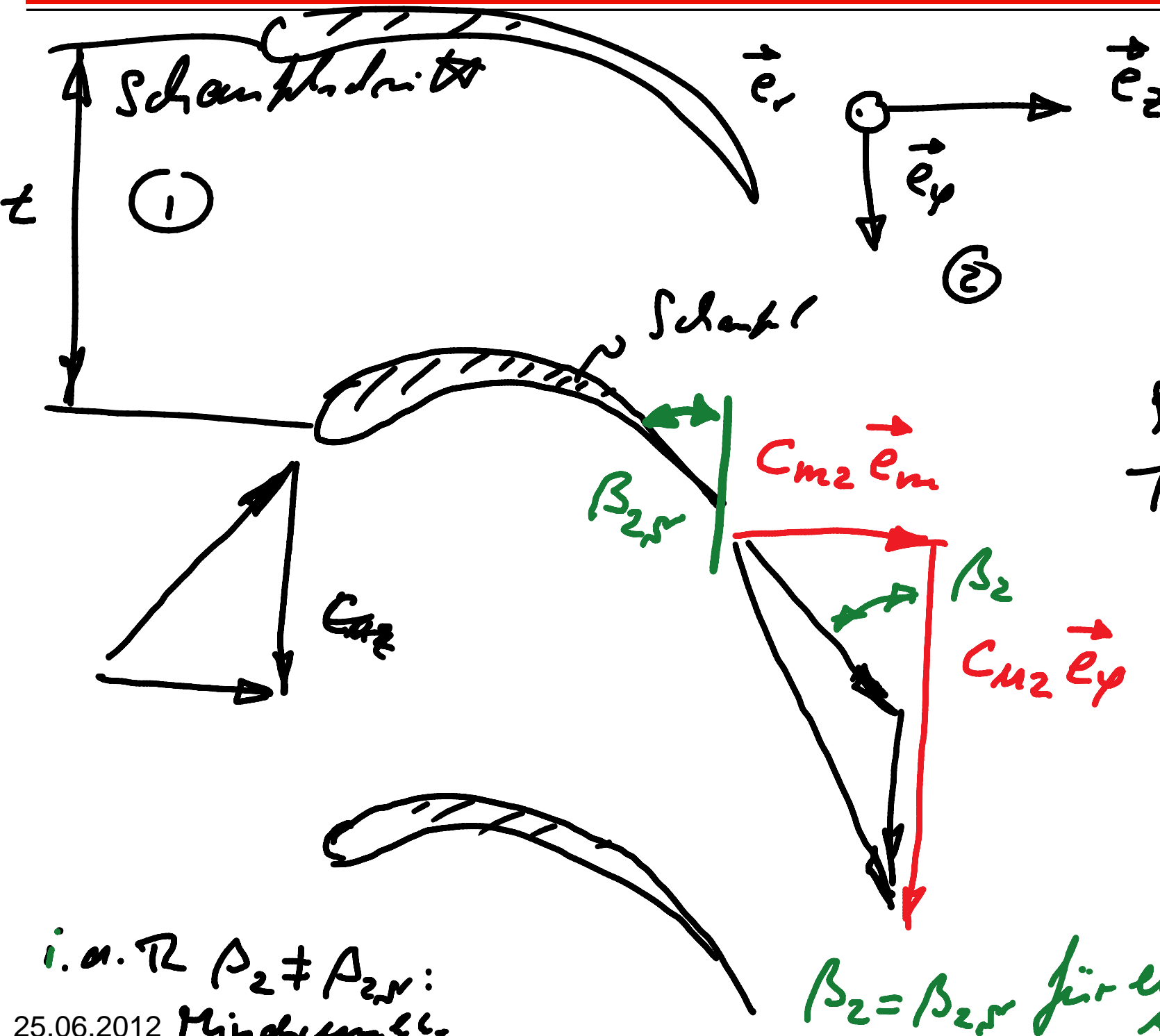
$$\frac{c_{u2}}{u_2}$$



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Grad, Schaufelbreite
 Teilung t
 "m" Meridianvill.
 $m=2$ bei
 ein Axialm.
 $m=r$ bei
 Radialm.

i.a. $\beta_2 \neq \beta_{2,r}$:
 25.06.2012 Mindermüll.

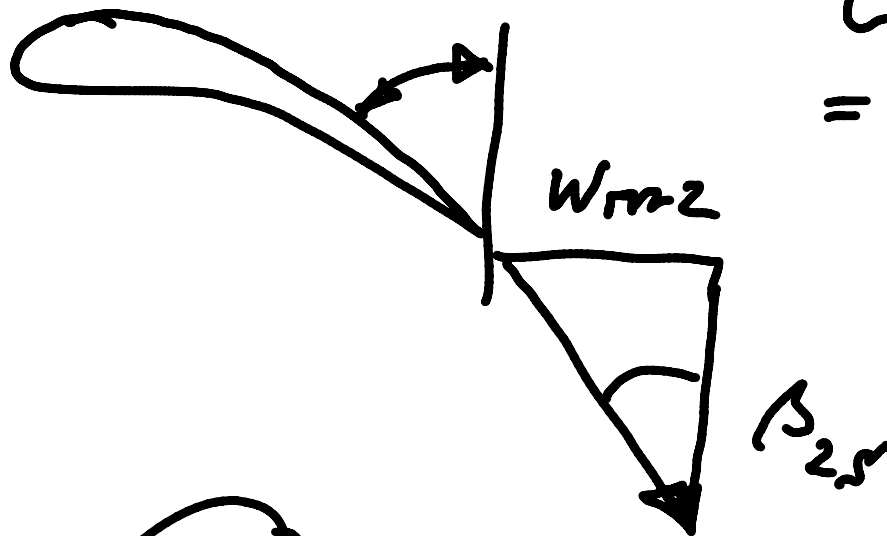


$$C_{M2} = \vec{c}_2 \cdot \vec{e}_y = (\vec{w}_2 + \vec{u}_2) \cdot \vec{e}_y$$

$$= w_{M2} + u_2$$

$$= \text{ctg } \beta_{2s} \frac{w_{M2}}{u_2} + u_2 \quad \Bigg| \quad \frac{1}{u_2}$$

$$= C_{M2}$$



$$\frac{C_{M2}}{u_2} = \left(\frac{C_{M2}}{u_2} \right) \text{ctg } \beta_{2s} + 1$$

φ Durchströmung

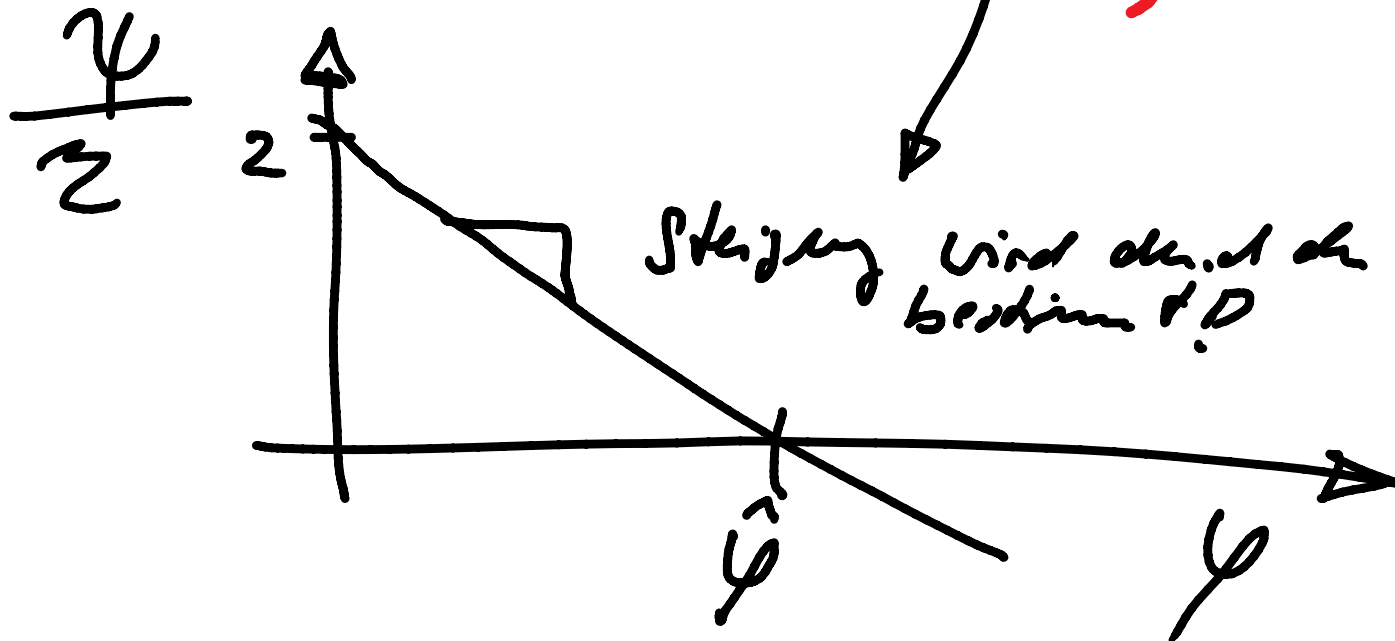
Einschleusen in die Kompressorlinie

$$\psi = 2z^{\pm 1} (-\gamma c_h \rho_2 + 1)$$

$$= 2z^{\pm 1} (1 - \gamma c_h \rho_2)$$

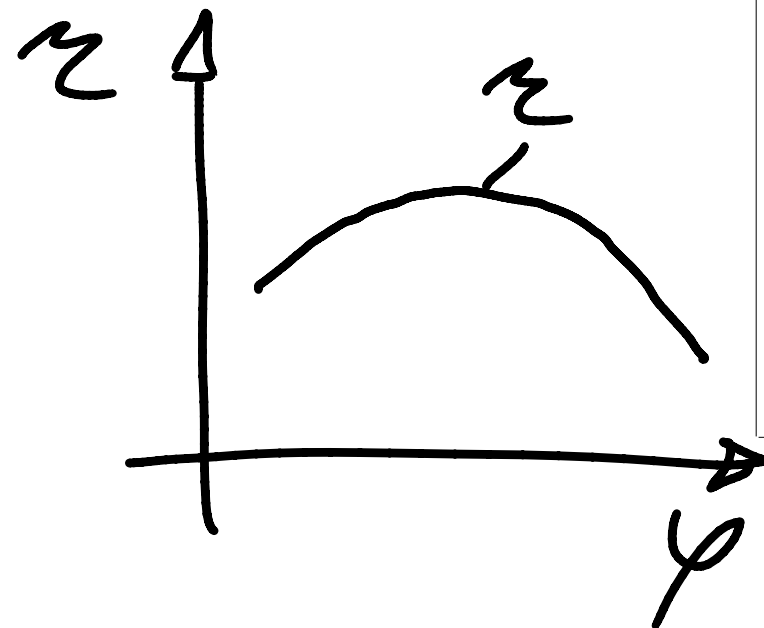
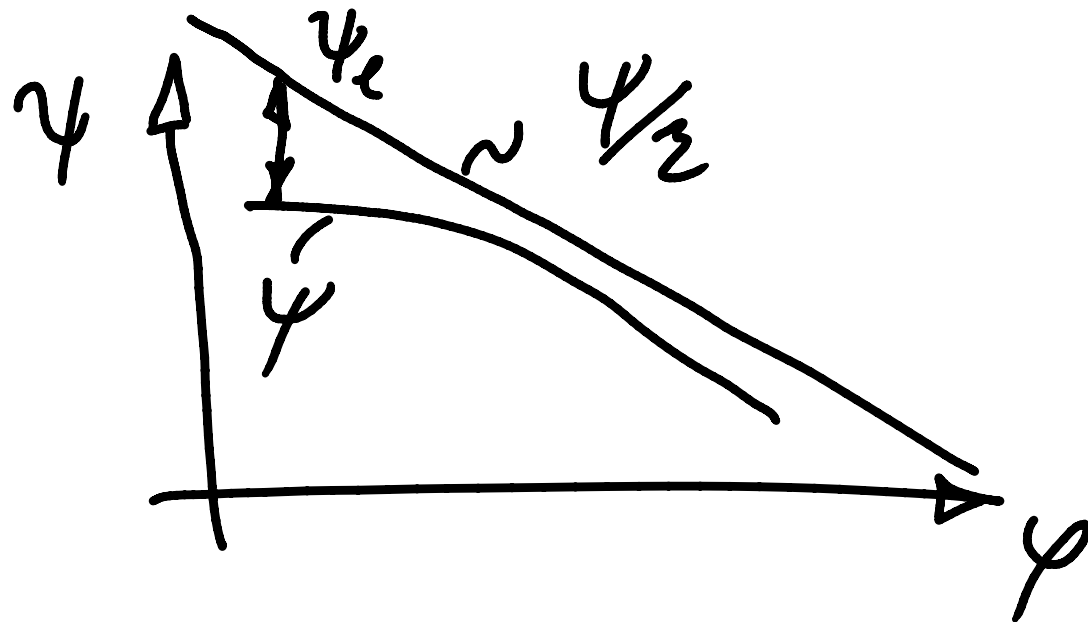
$$\frac{\psi}{z^{\pm 1}} = \psi_{\text{theoret}} = 2(1 - \gamma c_h \rho_2)$$

ideale Eulerkurve



Steigung wird durch die Machlinie bestimmt

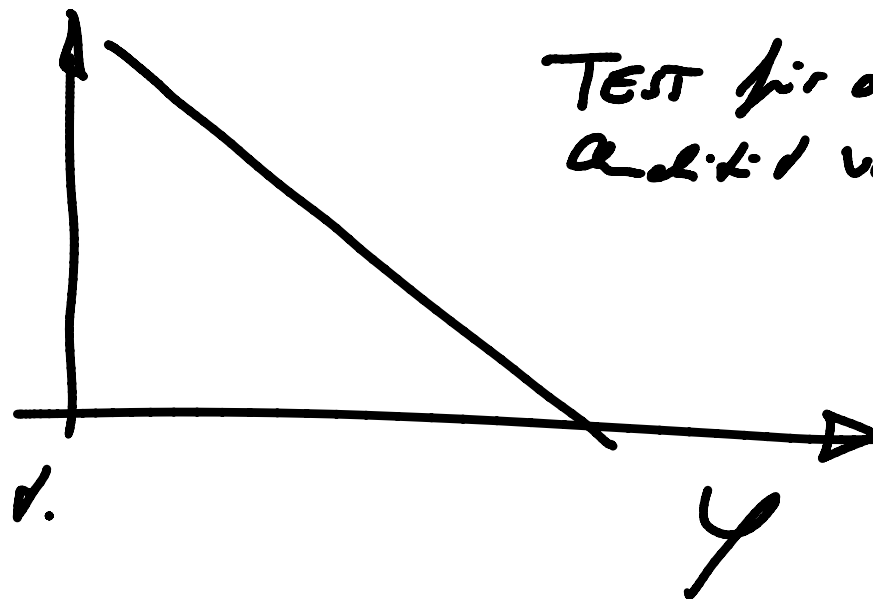




$$\frac{\psi}{z}$$

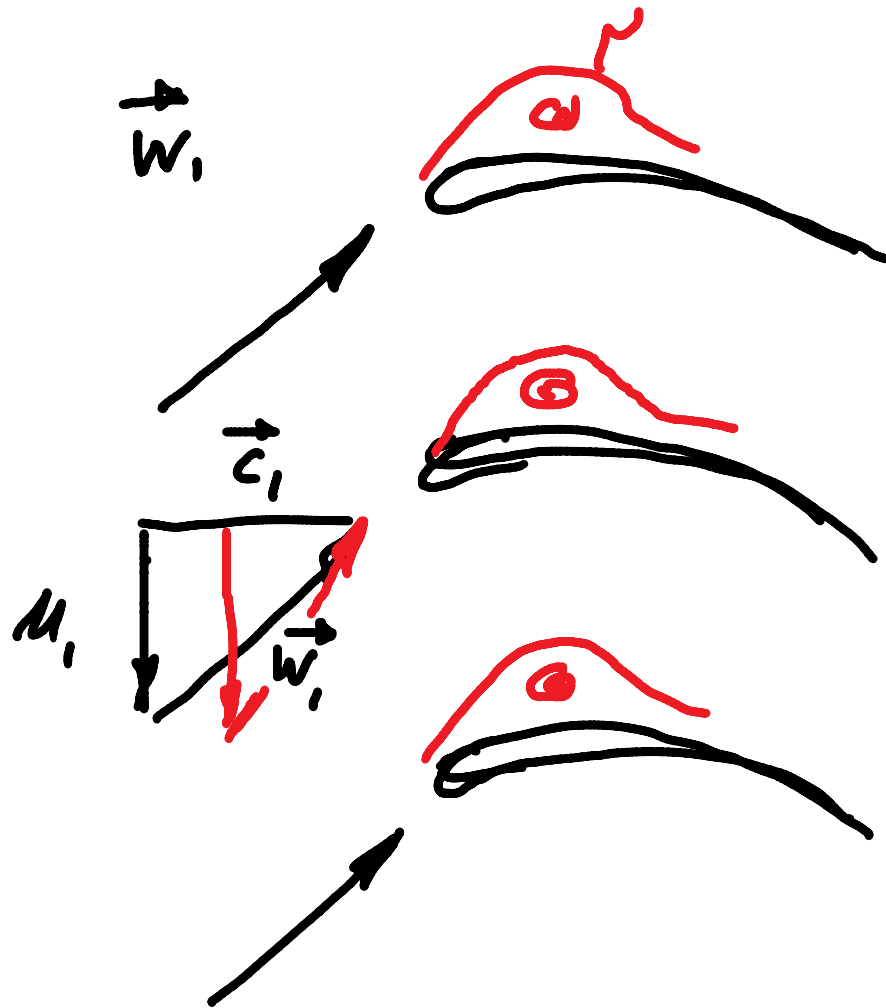
TEST für die
Qualität von Rotorbl.

ψ_e immer konstant.

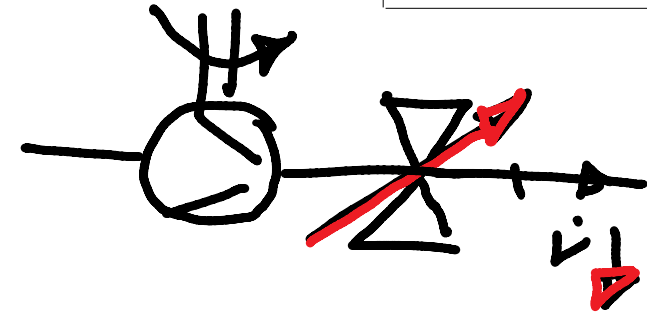


Wohin kommt die Drehleistung?

Teilverluste



$$h = c_{\text{aus}} \rho$$



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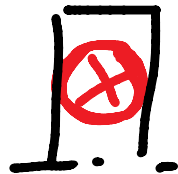
Der Analyseplan.



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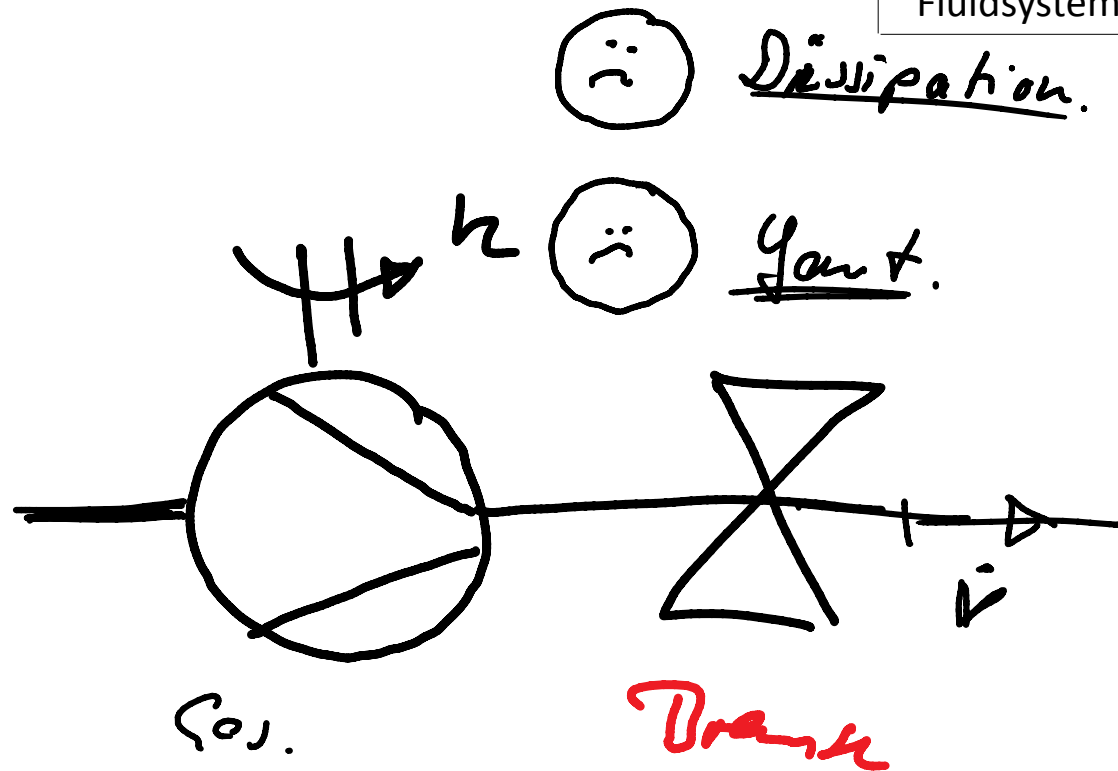
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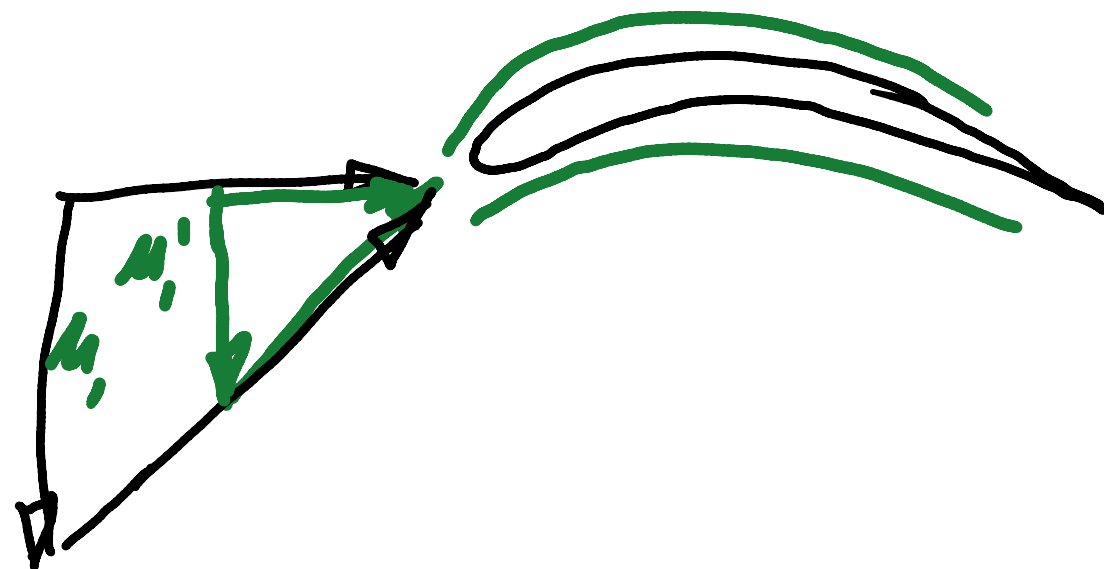
Druck



ζ₀₁.

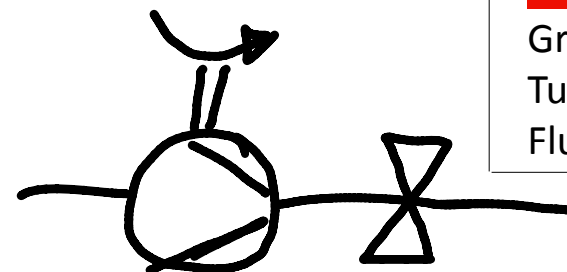


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$$h_1' < h_1$$

$$h_1 \rightarrow h_1' < h_1$$



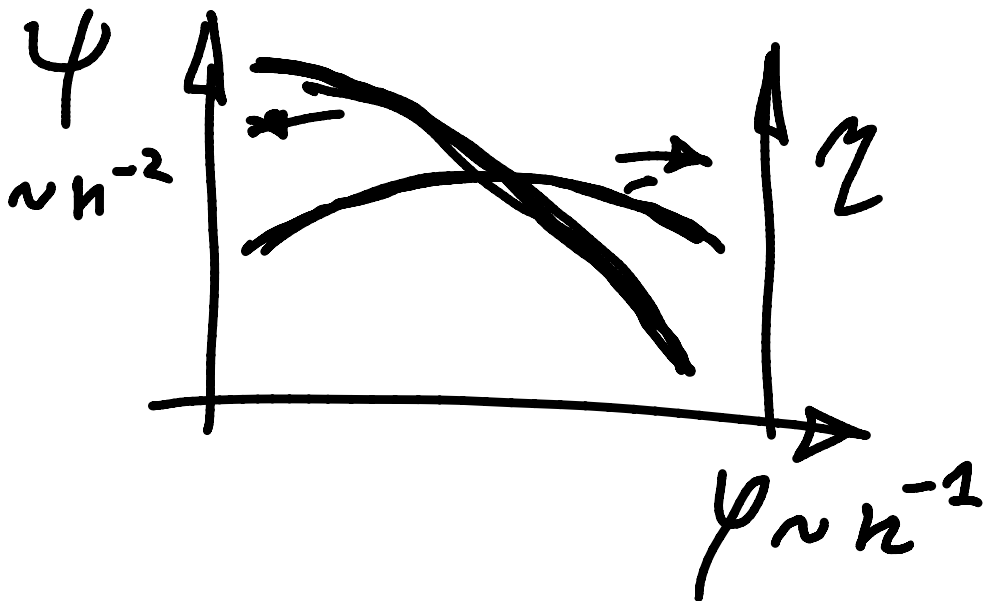
Druckabnahme!

Freigewandene Richtung → Rotor → Stator.



$$\psi := \frac{gH}{u_2^2/2} = \frac{gH}{\frac{2}{2} \frac{u_2^2}{2}} = \frac{gH}{u_2^2/2}$$

$$\psi := \frac{c_{m2}}{u_2} = \frac{\dot{V}}{A_2 \pi n d}$$



Kernlinie



Kernfeld

